## Exercise 7

- 1. Problem 16.5 in the course book
- 2. Problems 16.8 and 16.9 in the course book. For each of these problems plot on the same figure: 1) Bode diagram of WP; 2) Bode diagram of the open loop  $PK_{\infty}W$ .

(The notation  $b_{opt}$  in the book is parallel to the notation  $\alpha_{max}$  in the lecture slides)

3. Consider a simplified model of a satellite with two highly flexible solar arrays (Salehi, 10th IFAC Symposium, 1985)

$$\dot{x} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & 0 & -\omega^2 & -2\zeta\omega \end{bmatrix} x + \begin{bmatrix} 0 \\ 1.732 \cdot 10^{-5} \\ 0 \\ 3.786 \cdot 10^{-4} \end{bmatrix} u + \begin{bmatrix} 0 \\ 1.732 \cdot 10^{-5} \\ 0 \\ 3.786 \cdot 10^{-4} \end{bmatrix} v$$

where  $\omega = 1.539 \text{ rad/sec}$  is the frequency of the flexure mode and  $\zeta = 0.003$  is the flexural damping ratio. Here u is the control torque (Nm), v is a constant disturbance torque (Nm) and y is the roll angle measurement (rad).

Required performance specification: y must stay within  $0.04^{\circ} \approx 0.0007$  rad pointing accuracy due to 0.3 Nm step torque disturbances.

- Using [Zhou,16.12] prove the following bound

$$\left| (I - PK)^{-1}P \right| \le \frac{\gamma}{|W|}$$

where  $W = W_1 W_2$  is a single shaping function (for SISO plants).

- Show that the performance specification is met for the constant shape function W = 2000.
- Find the controller by  $H^{\infty}$  loop shaping procedure. Draw the Bode plots of the nominal P, shaped  $P_s$  and achieved open loop. Compare.
- Due to the steady-state error ( $\approx 0.018^\circ)$  introduce another shaping function

$$W = \frac{10000(s+0.4)}{s}.$$

Show that the performance specification is met and the full steadystate disturbance rejection is guaranteed.

- Repeat the third item for the new W.
- 4. Problem 16.3 in the course book. (This is an optional problem for those who are interested in detailed understanding of the derivations in Chapter 16.)